

Energy issues in WSN for Aeronautics Applications: Harvesting and Scavenging, Power Management, Storage

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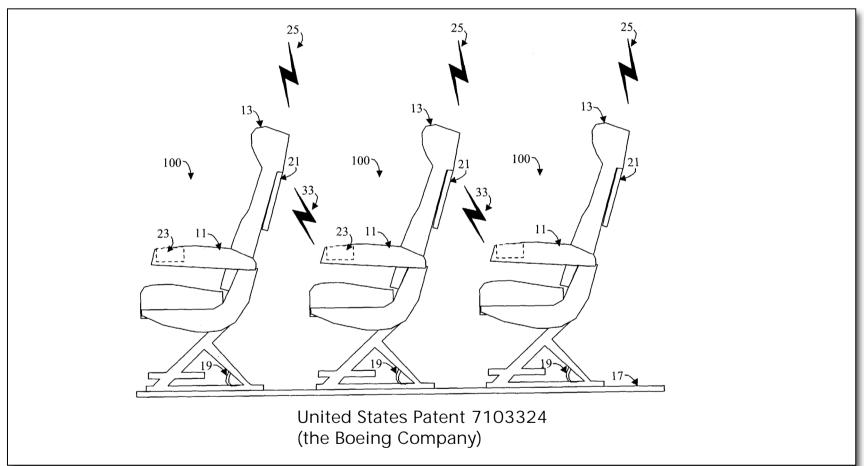


- Introduction: WSN and aircrafts
- WSN and SHM
- Energy issue: harvesting vs scavenging
- An example of energy capture: thermoelectricity
- Energy storage
- Energy management
- Network related considerations
- Conclusions



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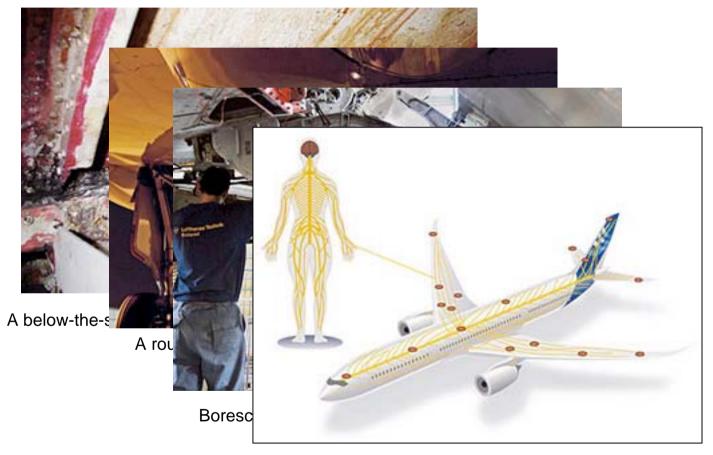




WSN for collecting passenger inputs (from passenger control units ("PCU") which are typically embedded in the armrest of the passenger seats)



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Level 1 (Detection).

Level 2 (Localisation).

Level 3 (Assessment).

Level 4 (Prediction).

SHM (EADS)

SHM: Structural Health Monitoring - sensors, some embedded in the airframe, detect cracks, corrosion, delamination and other damages and simplify their assessment while speeding their detection.



Def Stan 25-24 (UK MoD, 2004) defines health monitoring as "the automatic acquisition of data necessary to determine the potential failure or degradation of a system", which is echoed in FAA policy (FAA, 1999) where a health monitoring system is defined as "equipment, techniques, and/or procedures by which selected incipient failure or degradation can be determined"

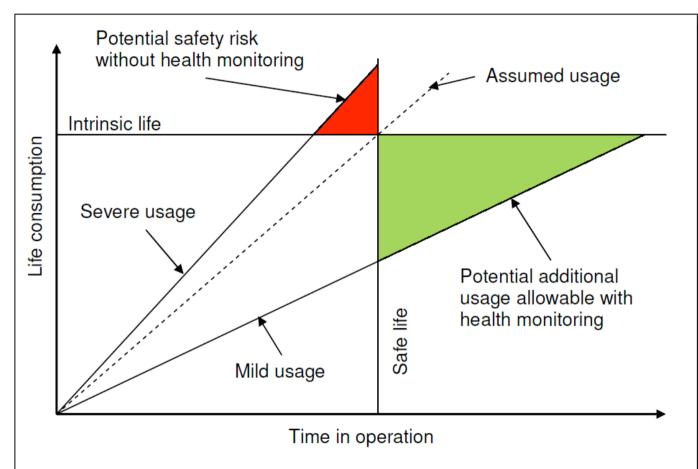
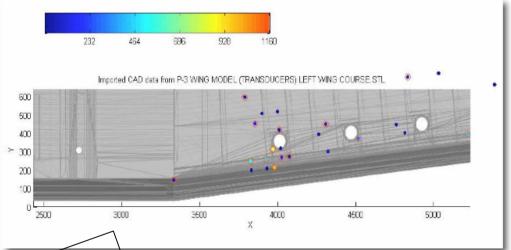


Figure 1-1 Potential Economic and Safety Benefits of Health Monitoring for Safe Life Components

(Romero, Summers, & Cronkhite, Feasibility Study of a Rotocraft Health and Usage Monitoring System (HUMS): Results of Operator's Evaluation, 1996)

Orion P3 - tanker







Acoustic Airframe Integrity Monitoring System (similar to that used for A340 fatigue tests: Balrue system)

(Aero Union Corporation / Ultra Electronics Ltd)



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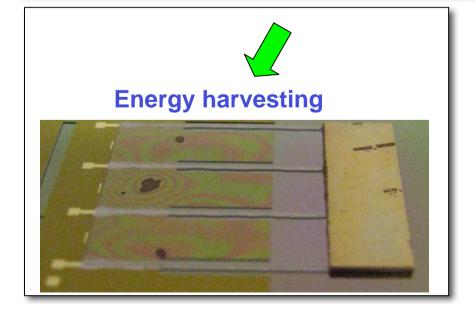
<u>Limited energetic autonomy:</u>

application is limited in duration sensors/nodes are accessible pollution issue if abandonned

<u>Unlimited energetic autonomy:</u>

long lifetime

'lost sensors' (or unaccessible) => energy capture from the environment





Energy scavenging

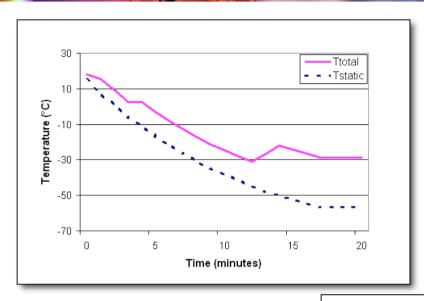


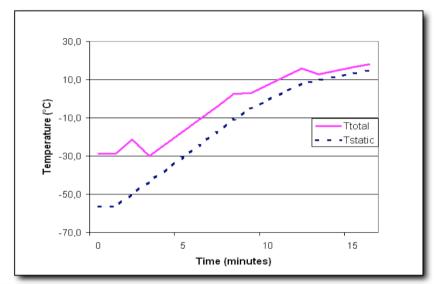
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Thermo-electric generation





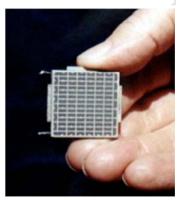


$$T_t = (1 + 0.2 \text{ M}^2) T_s$$



$$W_T = \Delta T \times C + H_f = 645 \text{ J/g}$$





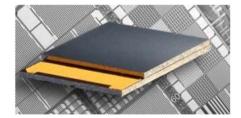


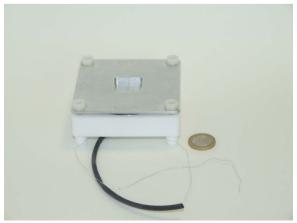








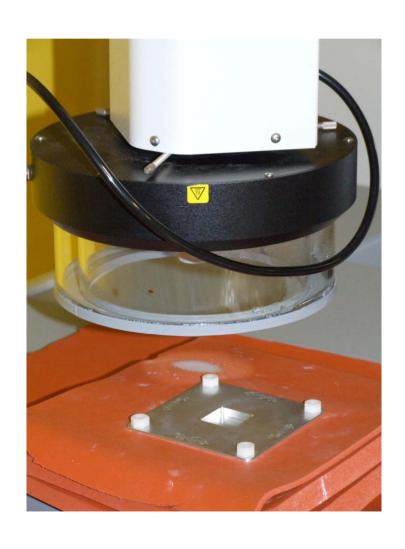


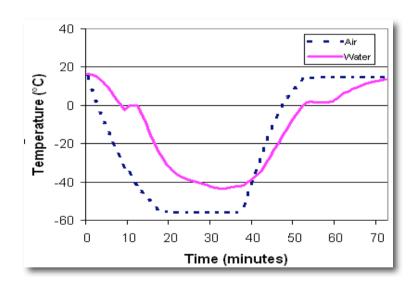










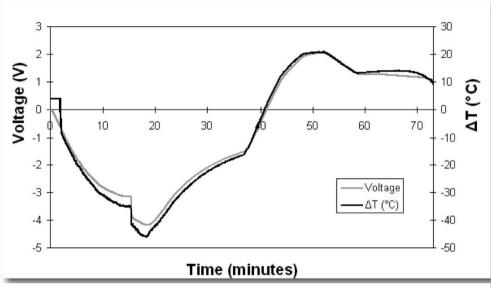


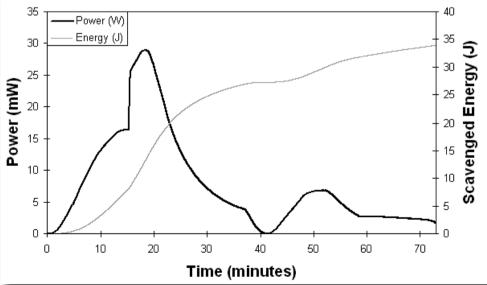


Thermal Gradients





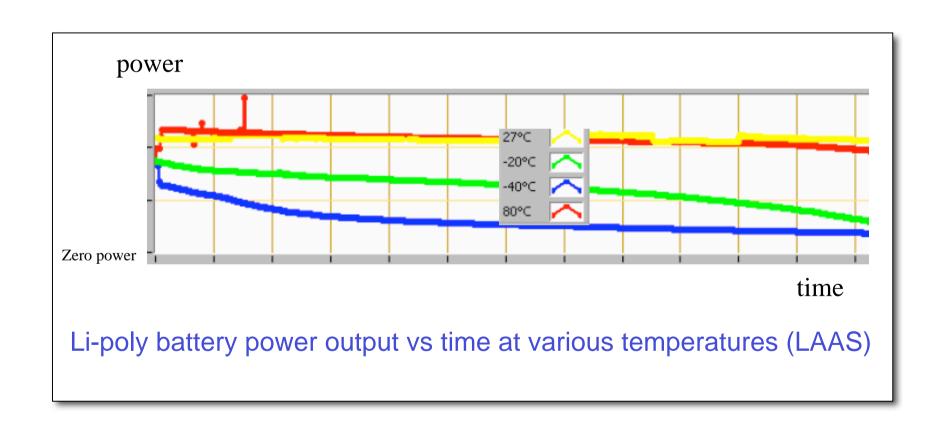


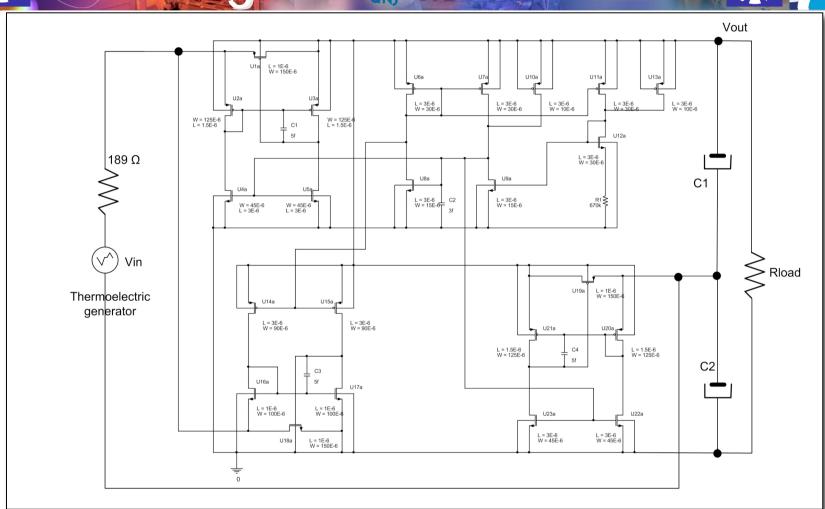




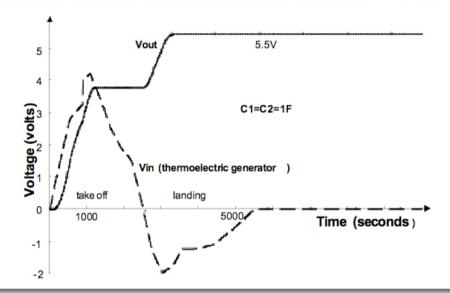
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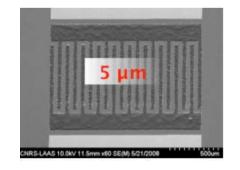


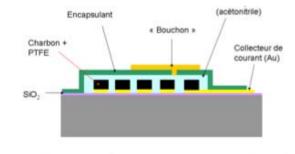


SPICE Simulation (0.35 µm technology): Rectifier threshold voltage = 10 mV Total Bias Current = 530 nA



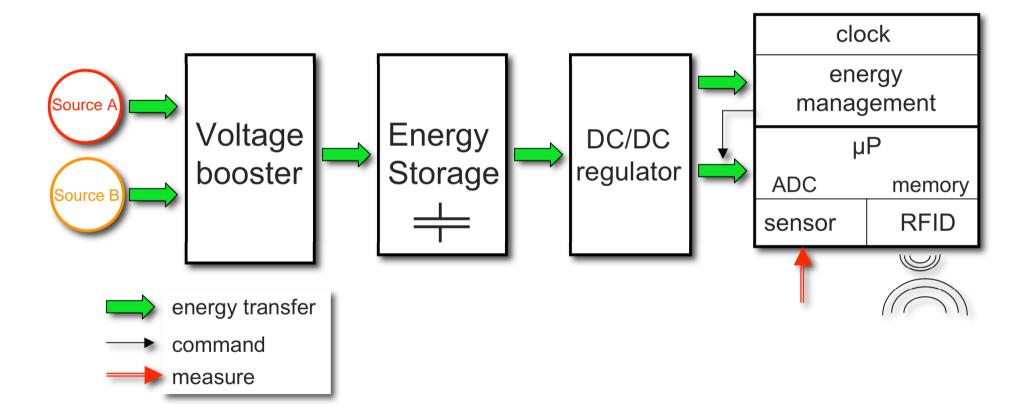
CMOS 0.35 µm
Active diodes Vt=10 mV
Self-bias activated at t=2min
DC/DC activated at t=9min (1V output)
Total bias current is 530 nA





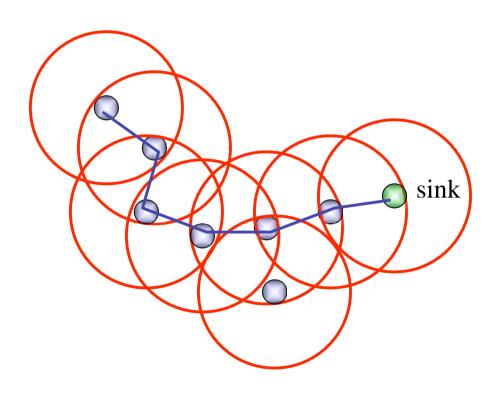


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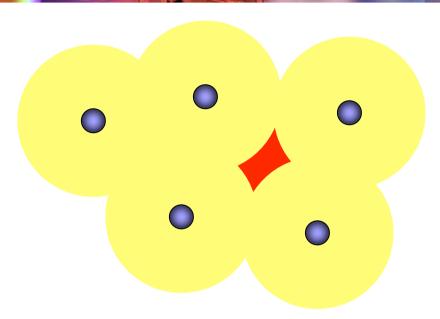
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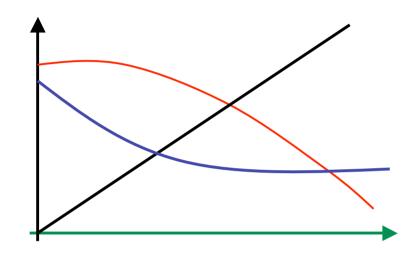




Network related issues:

- > none of the connectivity problems associated with multi-hop transmission found in WSN...
- > ... but a different approach no more based on the spatial distribution of nodes and their <u>transmission</u> range, but the spatial distribution of nodes and their <u>measurement</u> range





what is the relationship between

- sensor distribution density,
- captured power per sensor,
- sensor power consumption,
- total weight of the network?

... with the added perspective of short and long term resilience of network.



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